

High Technology Systems with Low Technology Failures

Some Experience with Rockets on Software Quality & Integration

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Launch Vehicles (aka Rockets)



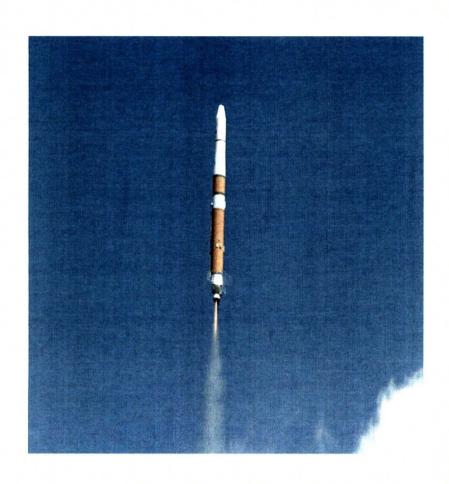


Background

- What is a launch vehicle (rocket) and what is it supposed to do?
 - A system which provides a spacecraft with the correct velocity and position at the correct time to be in the desired orbit
 - It just keeps adding velocity by accelerating until it gets to the desired orbit velocity
- Where did rockets come from?
 - The Chinese invented gunpowder and put it in bamboo tubes for essentially fireworks for festivals in about the 1st century AD
- Why did this technology not advance very far until the 20th century AD?
 - Inertial Guidance which consists of Navigation, Guidance & Control
 - This is the major functionality difference which separates model rockets from real rockets



Model Rockets vs. Real Rockets



Model Rockets \$5.00 - \$200.00



Real Rockets \$30 M - \$500 M



What is Navigation, Guidance & Control

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- Navigation Where am I?
- Guidance What is the path I want to take to get to where I want to go?
- Control How do I keep on the path to get there?
- The invention of the accelerometer and gyroscopes allowed us to measure the value and direction of acceleration
- Numerically integrate acceleration and you get velocity
- Numerically integrate again and you get position
- If you know where you started you can guide to your desired end point
- BUT it took a computer to perform the numerical computations



Quiz

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Question #1

What do you get when you cross a computer with a battleship?

Question #2

What do you get when you cross a computer with a camera?

Question#3

What do you get when you cross a computer with an automobile

Question #4

What do you get when you cross a computer with a rocket?



System Aspects

- Anytime you introduce computer technology into an engineered system you inherit the failure modes of the computer including software
- What is a system?
 - A construct or collection of different elements that together produce results not obtainable by the elements alone
 - Rockets have system requirements to accomplish their function as a system
- What is quality?
 - Conformance to requirements (does it work at the system level?)
 - Sometimes confused with luxury which is the presence or absence of certain product requirements or features



History of Rocket Failure Causes

- Early Flights
 - Design & Environment Errors
- More Recent Flights
 - Undetected processing errors (damaged systems)
 - Systems integration errors (systems do not play together like they should)
 - Collateral damage (propagated failures in complex systems)
- Some Rather Notable Failure Case Studies in Rocket Software Integration
 - Commercial Titan 3 Flight 2 (CT 2)
 - Ariane 5 Flight 1 (501)
 - Titan IV Centaur Flight 14 (TC 14)



Commercial Titan 3 Launch Vehicle





Commercial Titan 3 Flight 2 (CT 2)

- Commercial Version Modification of DoD/NASA Titan
 - Intended to compete in the commercial communication satellite market
 - Single and Dual Spacecraft Payload Configurations
- First Flight was a Dual Payload Configuration
 - Successfully Injected and Separated the Skynet 4A & JCSAT2 payloads on December 31, 1989
- Second Flight was a Single Payload Configuration
 - Failed to Separate the Single INTELSAT 603 Payload when commanded when launched on March 14, 1990
 - \$265 M Loss (Uninsured \$150 M Payload)
- Incorrect Software/Electrical Configuration when switching from Dual to Single Payload Configuration
 - Inadequate testing to detect the design error
- CT 3rd flight with the same INTELSAT 6 payload was a success



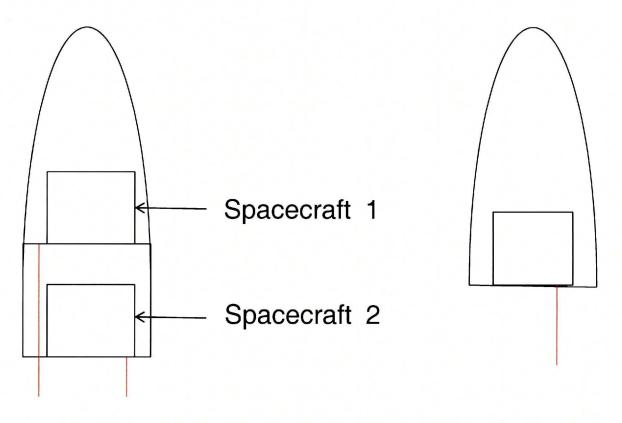
Commercial Titan 3 Payload Configurations

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Dual Payload Configuration

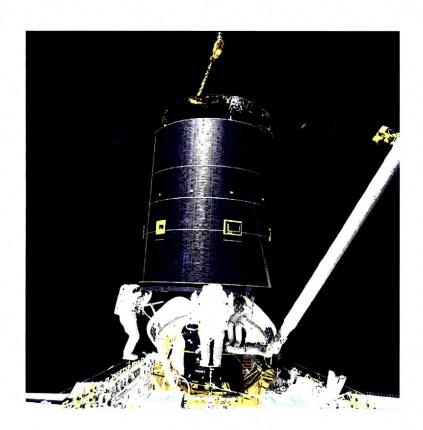
Single Payload Configuration

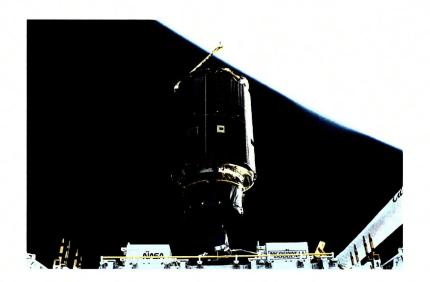


Separation Command Wire Harnesses



The Rest of the INTELSAT 603 Story



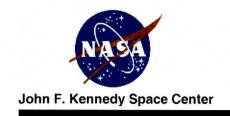


STS - 49 Retrieval Mission



Ariane 5 Launch Vehicle





Ariane 501

- First Flight of a major upgrade from Ariane 4 to Ariane 5
 - Addition of large solid rocket boosters
 - Intended to capture a large share of the commercial communication satellite launch market
 - \$ 7 Billion development program
 - Launched on June 4, 1996
- 37 seconds after engine ignition the vehicle abruptly changed attitude, broke apart due to aerodynamic forces and initiated the vehicle flight termination system
 - \$370 M Loss
- Control system sent an unneeded major attitude change signal to the engine control actuators
- Guidance system sent erroneous data to the control system due to a shutdown of the inertial measurement unit (IMU) computer



Ariane 501 Flight



Liftoff!



37 seconds later !!!



What happened to Ariane 501?

- Data transmitted from the IMU was not proper flight measurement data but was a diagnostic bit pattern of the IMU computer which was interpreted as flight data
- The IMU computer had declared a failure due to a software exception
- The internal IMU software exception was caused during execution of a data conversion from 64 bit floating point to a 16 bit signed integer value
 - The floating point number which was converted had a value greater than what could be represented by a 16 bit signed integer
 - Related to a horizontal velocity measurement
 - Resulted in an Operand Error
- The data conversion instructions in Ada code were not protected from causing an Operand Error



What happened to Ariane 501? (cont)

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- The exception handling mechanism for the IMU computer was
 - The failure was to be indicated on the databus
 - Failure context stored in EEPROM
 - Processor shut down
- Attempted to switch to a redundant IMU processor
 - Could not do so because the redundant IMU processor had failed during the previous data cycle for the same reason
- Error occurred during in a part of the software which is used for strap down inertial platform alignment
 - Provides meaningful results only prior to liftoff
 - After liftoff this function serves no purpose
- The alignment function was operative for 47 sec after liftoff
 - Time sequence was based on requirements for Ariane 4 and was not required for Ariane 5 (Reuse/commonality of software)
 - Used for rapid realignment of IMU on Ariane 4



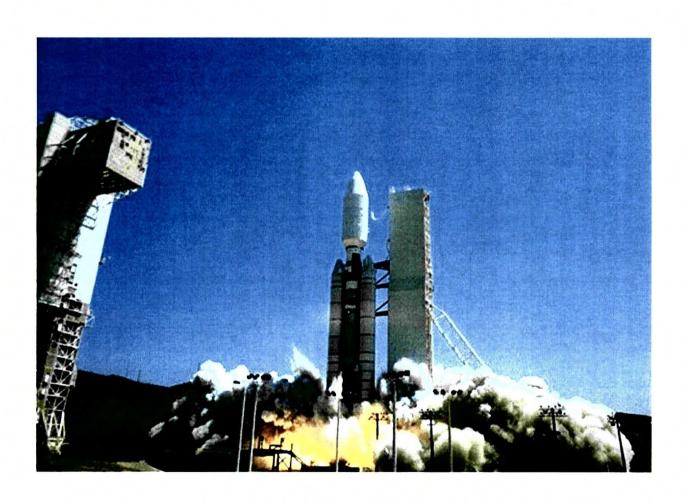
What happened to Ariane 501? (cont)

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- The Operand Error occurred due to an unexpected high value of a variable called horizontal bias
 - Related to sensed horizontal velocity
 - Horizontal velocity value for Ariane 5 was about 5 times the value for Ariane 4
- IMU System was not tested with the simulated Ariane 5 trajectory
 - IMU specification did not contain the Ariane 5 trajectory as a functional requirement
 - When they did so the failure was duplicated
 - IMU specifications did not indicate operational restrictions



Titan IV Centaur Launch Vehicle





Titan IV Centaur Flight 14

- Titan IV Centaur is a US launch vehicle used for DoD & NASA missions
- Launched on April 30, 1999
- Upper stage (Centaur) tumbled out of control after spinning itself at a value an order of magnitude too high
 - Spacecraft placed in useless orbit
 - Due to an incorrect roll sensor gain software parameter value off by an order of magnitude (or one decimal place)
 - \$ 1 Billion Payload Loss (Uninsured US Air Force payload)
- Roll is motion about the longitudinal axis of a rocket
 - Occurs while sitting on the earth due to earth rotation
- Incorrect software value could have been detected in prelaunch testing data analysis
 - In specification but out of family



Failure Mitigation Strategies

- Practice Systems Engineering
 - Know how everything works as a system
 - Have domain knowledge of functionality/criticality
 - In a control system everything matters (sensors, computation, actuators and their data)
 - Software has imbedded assumptions in its logic
- Think about what can go wrong
 - Success is eliminating/mitigating causes of failure
 - Maybe use some formal failure analysis techniques
 - » FMECA, Event Sequence Diagrams
 - Think about how to make systems robust
 - Study technology history and learn from others mistakes
- Devise simple sanity tests/data analysis to eliminate errors
 - Think functionality
 - Analyze the data
 - Isolate one function at a time



Failure Mitigation Strategies (cont)

- Testing
 - Tests to prove no possibility of a negative function
 - Graceful degradation/failure handling
 - Try to break the software
- Progressive levels of integration testing
 - Elements may work by themselves but not together
- End to End and Integrated Testing
 - Test Like You Fly
 - Fly Like You Test
- Random Hardware Failures are Rare
 - Design, Integration, Testing & Data Analysis Failures are NOT